2.6 <u>Est</u>

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Establish Time the Launch Site Crosses the Orbital Plane (Continued)

- $\Omega_{e}t$ = longitude of the prime meridian relative to T at the time the launch site crosses the orbital plane.
- $\Omega_{e}t = A_{L} A_{L}$ by inspection.

The geocentric latitude, L, may be obtained from the geodetic latitude, L', by means of

$$tan L = (1-f)^2 tan L'$$

1

The spherical triangle of interest shows the launch site at the instant of intersection with the plane of motion.



 $sin(A_L-\Omega) = tan L_L tan (90-i) \rightarrow (A_L-\Omega)$

$$= sin^{-1} (\underline{\qquad} tanL_L) \\ tan_1$$

(Note that the inclination must be equal to or greater than the latitude for the procedure to work.)